## Background

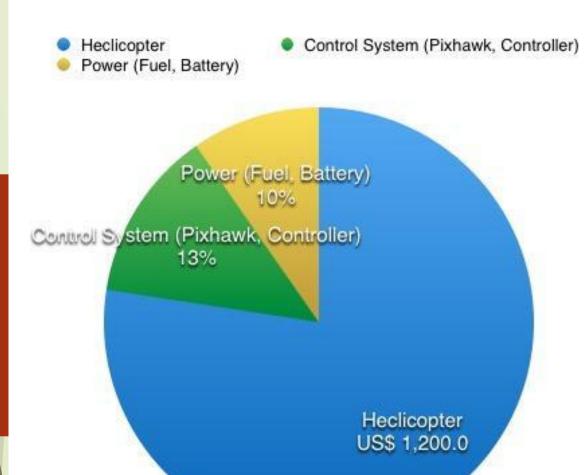
A helicopter is a rotor device, which can take off, land vertically, controlled by four operating controls. The collective pitch controller is to change to angle of attack on both blades for a uniform lift. The cyclic pitch which is responsible for the pitch and roll for the helicopter. The throttle that has a main purpose to control the angular speed of the main rotor Lastly, the anti torque control is used to cancel the main rotor torque and change the yaw angle.

# Goal & Objective

Our goal for the project is to engineer a remote controlled helicopter with auto piloting. Learn and examine the control system of a helicopter, which is different from a normal fixed wing aircraft.

Design and program the helicopter for a self regulating autonomous flight.

# Budget



### Team Member & Contact Info

Team Lead : Yunliang Sha, yunlians@uci.edu Safety Manager : Yizhou Pan, yizhoup1 @uci.edu Document Manager : Haoran Yu, haorany2@uci.edu Purchasing Manager : Runjung Li, runjingl@uci.edu Flight Engineer : Zihao Zou, zzou1@uci.edu

### Winter

- -Helicopter (prototype and small scale test) assembly
- -Circuit design & Remote control flight -Control system theories

### Spring

-Control system programming -Prototype test flight

# Helicopter DBF

# Advisor: Haithem E. Taha | Colin Slege





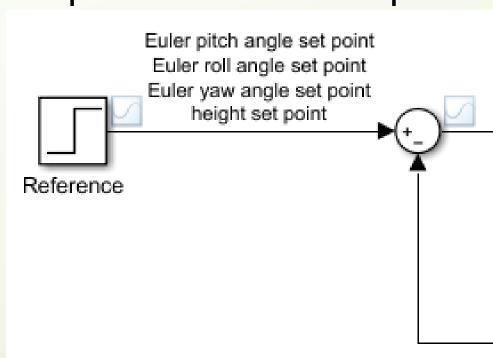
The Pixhawk will send the control input to the swash plate which will execute.

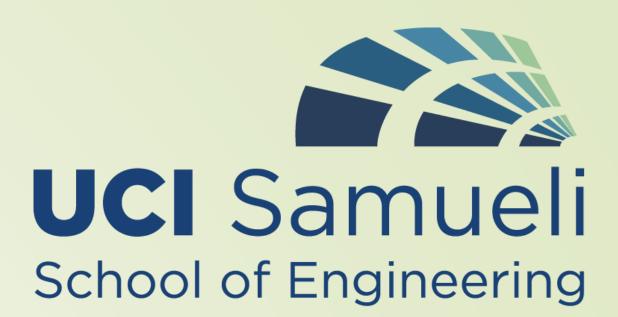


error

-Small scale implementation and data analysis

#### Simplified Control Loop





## Requirement

- RC aircraft able to carry around 5 kg of motor, servos, and sensors to conduct the flight as planned.
- Aerial computer system with sensors and communication devices to receive command, execute control inputs, and collect data.
- Ground computer system for flight missions and data collection.

## Innovation

- Self regulating flight control system with multiple functions.
- Fully automatic data collection and flight mission.

Our team is using Pixhawk to implement the designed control system

r	Contro	Cyclic lateral ontrol input Cyclic longitudinal control input Collective control input Antitorque control input	•	Input	Measured Output
	Pixhawk	-			Helicopter
		Euler pitch angle Euler roll angle Euler yaw angle height			